#### PRE-FLIGHT WIRING REVIEW EXTERNAL TANK PROJECT READINESS STATEMENT STS-103/ET-101

PRE-FLIGHT WIRING REVIEW DATED NOVEMBER 1, 1999, IS CONSIDERED READY TO EXTERNAL TANK ET-101/STS-103, AS IDENTIFIED IN THE EXTERNAL TANK PROJECT SUPPORT FLIGHT UPON ACCEPTABLE DISPOSITION OF OPEN/PLANNED WORK

OF J. BULLMAN/ED10

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#### STS-103/ET-101 Pre-Flight Review

#### Agenda

#### Overview

### Current Mission/External Tank Highlights

ET-101 Mass Properties Status

Changes

Special Topics

- STS-99/ET-92 Pre-Flight Review Action Item
- Critical Process Changes
- Weld Instruction Card (WIC) Certification
- ET Electrical Harness Investigation
- ET/SRB Cross-Strap Harness Removal/ Replacement

Senior Management Review NCDs: None

SRM&QA Assessment

KSC Processing

Verification/Certification

Mission Unique Assessment

Readiness Statement

**Appendices** 

- A: Minor Changes
- B: NCD Assessment Criteria
- C: KSC Processing
- Level II Exceptions/Waivers and Level III Deviations
- E: Items previously reviewed for STS-99/ET-92

#### Jim Feeley

Rick Spring

Jim Feeley

Don Bolstad Leed Colon

Mike Bankester Juan Ramirez Jim Feeley

#### Overview

### This review for STS-103/ET-101 includes:

- Items specific to mission STS-103
- First time changes, differences and "out-of-family" non-conformances not previously flown and/or reviewed
- STS-99/ET-92 previously scheduled to fly before STS-103/ET-101
- STS-99 Pre-Flight Review (9/1/99) included items applicable to ET-92 and ET-101
- Common items will not be re-addressed in this Pre-Flight Review
- Items are included in Appendix E, Items Previously Presented for STS-99

#### **FRR Series**

<ul> <li>Space St</li> </ul>	<ul> <li>Pre-Flight Review</li> </ul>	• ET/SRB
Space Shuttle Vehicle FRR	t Review	ET/SRB Mate Review
11/19/1999	11/08/1999	08/24/1999

Mission Management Simulation

Mission Management Team

11/18/1999

### Mission/External Tank Highlights

### Mission STS-103

#### Launch

- Window
- Pad
- Orbiter
- Orbital Inclination
- ET Photo Coverage

12/07/1999 (Under review)

TBD

Discovery (OV-103)

28.45° - Insertion at 317 NM

- Crew photos from cockpit window
- Umbilical well cameras
- SRB cameras

#### **Primary Payload**

#### Landing

- Time
- Location

#### 12/17/1999 (Under review) **Hubble Servicing Mission 3** TBD

#### **External Tank**

DD250 Acceptance

Shipped to Launch Site

#### ET-101

KSC

01/15/1999 11/25/1998

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### **ET-101 Mass Properties Status**

ET-101 Specification Weight = 58,470 lbs

ET-101 was weighed 02/08/1999 @ KSC

Predicted 58,337 lbsAs-weighed 58,465 lbs

hed 58,465 lbs +128 lbs

Delta

Level II uses "as-weighed" data for mission planning

ET-101 was weighed with compression platform scales at KSC

Suspect PR written on KSC weighing system due to scale platform interference

PR disposition in work that invalidates ET-101 actual weight

LMMSS recommends using predicted weight for mission planning

MMC-ET-SE40, ET Project - Mass Properties Weight and Balance Report, will be resubmitted to note that ET-101 as-weighed data is suspect (ECD 11/12/99)

for STS-103 Preliminary coordination with Level II indicates approximately 12,000 lb performance margin

No issues for mission planning

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#### Changes

- Response to MIL-Spec Cancellations (B02047)
- Roseann Gray
- LH2 Tank changes to accommodate SSME Block II Implementation (B02034)

Mike Quiggle

- Intertank thrust panel foam loss corrective actions Vent thrust panel foam (FEC KET-0054 and R1)
- Vent additional thrust panel foam and +Z skin/stringer panel foam (FEC KET-0056)

**Eugene Sweet** 

## Response to MIL-Spec Cancellations

#### Change

Created and implemented a document which lists alternate specifications for cancelled MIL-Specs

#### **Background**

- specifications, and standards (MIL-Specs) in design and manufacture Aerospace industry has historically used U.S. government, military and federal
- Government, military and federal specifications and standards U.S. government Acquisition Reform legislation has resulted in cancellation of U.S.
- ET engineering requirement documents (such as drawings, standards, engineering material and process specifications) specify over 350 MIL-Specs
- LMMSS reviewed engineering requirement documents for reference to MIL-Specs
- Significant effort would be required to revise all engineering requirement documents referencing the cancelled/inactive MIL-Specs
- LMMSS initiated activity to approve replacement specifications
- Material Sciences 'Senior Review' Committee established to review and approve replacement specifications for cancelled MIL-specs
- Weekly meetings conducted to review alternates for technical equivalence
- Specifications not considered equivalent will be processed through normal change processing system

## Response to MIL-Spec Cancellations

#### Description

- alternate specifications for cancelled/inactive MIL-Specs Created book form collector drawing (809-3600) which lists technically equivalent,
- Added drawing reference to top ET assembly level drawings and to Exhibit A for supplier documentation requirements
- Revise ET Materials and Processes Control Plan (SE-16) to add reference to 809-3600 as the collector document for identification of alternate specifications
- 'Senior Review Committee Updates to collector occur as potential alternates are approved by the Material Sciences

#### **Basis for Certification**

Similarity - Technical review of alternate specification is conducted to confirm that identified specifications are technically equivalent to the cancelled specifications

# LH2 Tank Changes for SSME Block II Implementation

#### Citalige

- Increased pre-launch ullage pressure requirement to accommodate SSME Block II Implementation
- ET GH2 vent/relief valve requirements revised for higher pre-press level
- No design change required
- Reseat pressure
- Was: 34.0 psig minimum
- Now: 34.25 psig minimum
- Acceptance test instrumentation and requirements revised
- Perform post proof weld x-ray requirements on additional 215 inches of under proofed LH2 tank circumferential welds

#### Reason

- temperature spikes during engine start transient causing reduced turbine blade life Advanced high pressure fuel turbopump in the Block II SSME has increased preburner
- Required modification of the ET pre-pressurization control bands and LH2 ullage pressure ICD
- Raised pre-pressurization control band will be used only on flights with three Block II
- Not required for STS-103/ET-101

# LH2 Tank Changes for SSME Block II Implementation

### **Basis for Certification**

#### Test and Inspection

- Successful completion of LH2 tank proof test and post-proof inspection
- 215 inches of additional weld inspected
- No change to critical test demonstrated LH2 tank Factor of Safety
- Raised LH2 tank pre-press band was demonstrated on STS-91 tanking test
- Narrow band with 0.5 second GHe bursts was demonstrated
- Pre-press level demonstrated was 0.3 psi lower than planned for use with three **Block II SSMEs**

#### **Analysis**

- Propulsion analysis shows LH2 tank ullage pressure will be within LCC limits during pre-press
- Structural analysis shows overall critical factor of safety unchanged
- Factor of Safety for critical circumferential welds (failure mode: ultimate tension)

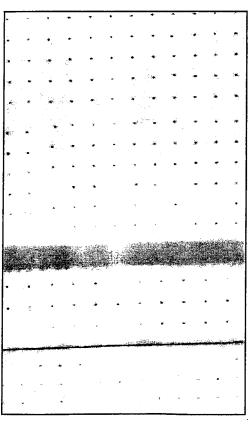
• LH2 Tank H5	<ul> <li>LH2 Tank H4</li> </ul>	Weld ID
1.25	1.25	Required
1.49 1.42	1.51 1.45	Was Now
1.25 1.25	1.24 1.25	Proof Test Demonstrated Was Now

#### Change

- Vent portions of the Intertank thrust panel and skin/stringer panel foam
- 0.032" diameter pin holes are spaced 0.30" apart

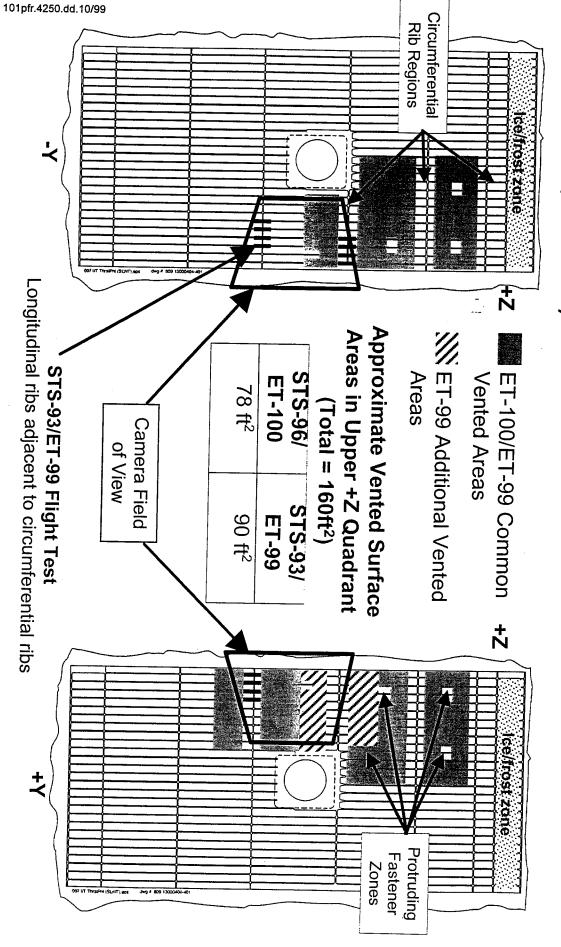
#### Background

- by foam loss from the ET Intertank thrust panel Post flight inspection of STS-87 revealed out-of-family damage to the Orbiter tiles caused
- popcorn-type debris A rigorous test program has demonstrated the potential for vented foam to reduce
- Vented foam configuration has been certified by test and analysis to do no harm
- Venting of Intertank foam implemented on STS-96/ET-100 and STS-93/ET-99
   Review of SRB video following STS-96/ET-100 showed less debris with vented foam
- Based upon STS-96/ET-100 data and additional analysis, area of Intertank thrust panel foam to be vented was increased for STS-93/ET-99

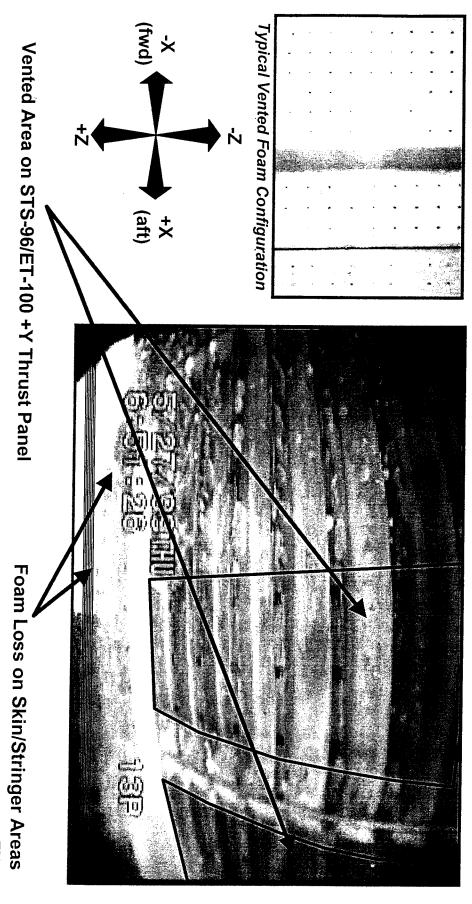


Typical Vented Foam Configuration

### Background (continued)



- Background (continued)
- Review of STS-93/ET-99 SRB flight video confirmed performance enhancement realized through foam venting
- Popcorning exhibited on Intertank skin/stringer foam

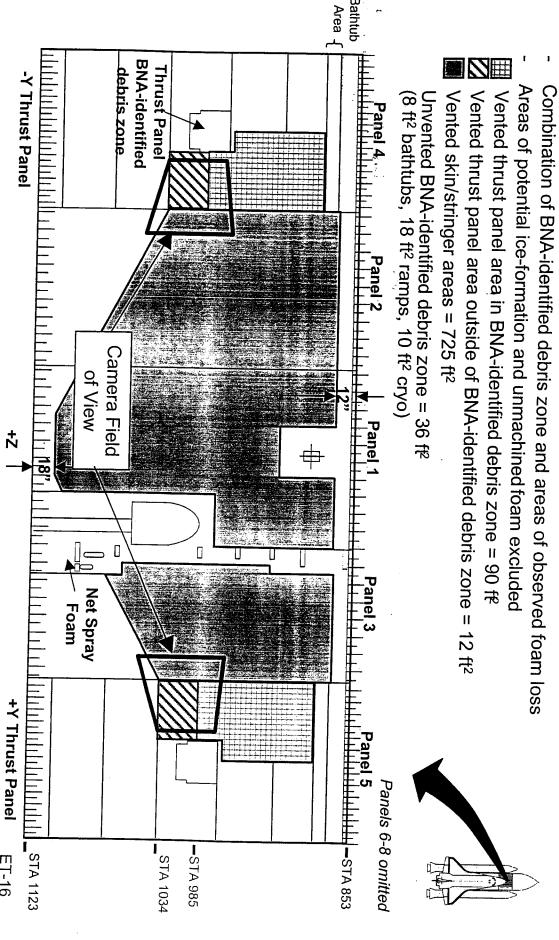


ET-16

# Intertank Thrust Panel Foam Loss Corrective Actions

### **Description of Change**

- Perform venting of Intertank foam areas



### **Basis for Certification**

- Primary test facilities used were MSFC Hot Gas and Thermal/Vacuum--the best discriminators of popcorn performance
- Popcorning performance of additional rib and skin/stringer configurations similar to that of previously successfully vented thrust panel configurations
- certified "to do no harm" locations adjacent to circumferential rib ramps and skin/stringer panel areas are NASA/LMMSS IFA team reviewed test results indicates that venting of rib
- No safety of flight concerns

### Basis for Certification Test

- Tests performed in different test beds following various environmental conditioning
- Results from all performance testing show that vented foam performs as well as or better than the nonvented foam configuration and measurably reduces foam loss
- Similarity
- Vented foam configuration similar to configuration flown on STS-96/ET-100 and STS-93/ET-99

	Certification Test	No. of			
		Tests	riuminuty	Salt Fog	Vented
	veilled Foam Certification Testing				
	Mechanical Properties/Acceptance Testing				
	Density	200			
	Bond Tension	F 40	<		×
	Flatwise Tension	140	: >	×	×
	l an Shear	040	×	×	×
	ריקל כווכמו	360	×		×
	Plug Pulls	144		<b>&lt;</b>	< >
5	Flexure (Ribbed panels)	24	×	>	< ×
=	Thermal Properties				×
	Thermal Conductivity	24	<	< 	<
	Flight Verification	!	>	>	<b>×</b>
	Hot Gas - Flat Panels, Machined foam	20	<	<	(
	Hot Gas - Rib Panels, Machined foam	6 C	< >	< <i>&gt;</i>	: ×
	Hot Gas - Skin/stringer, Machined foam	۰ S	< >	×	×
	Thermal/Vacuum Elat Danala Maaking Com	; `	>		×
	Thermal/Vacuum Size - Bit 5	40	×	×	×
	The It I action of Silver Rib Panels, Machined foam	. 14			×
	Thermal/Vacuum, Rib Panels, Machined foam	24	×	×	× ;
	Merinal/Vacuum, Skin/stringer, Machined foam	8	×		×
	wind lunnel (AEDC), Machined foam	4			× :
	Vibro/Acoustic Test (DOE C Addendum Testing)	<u>ი</u>	× 		< >
	Process Verification/Acceptance		:		>
	Full-Scale Process Pathfinder (GVTA)	<u>-</u>			<
					>

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#### **Action Item**

critical process changes are controlled Develop and document criteria for critical processes and present how

#### Response

Definition of critical processes is documented in NSTS 5300.4 (1D-2) where design conformance cannot be assured by inspection. A material process (I.e., a process which changes the chemical/ performance effect on hardware identified on the CIL (Critical Items physical properties of a material) which could have a significant List), hardware designated for fracture control and ordnance, and

#### Response

- 5300.4 External Tank critical processes are consistent with critical process definition in NSTS
- Forming & Fabrication of Metals
- Forming & Fabrication of Non-metals
- Fasteners
- Non-Destructive Evaluation
- Finishes
- Testing
- Lubrication
- Identification & Packaging
- Cleaning & Contamination Control
- Welding, Brazing & Soldering
- Adhesive/Bonding
- Electrical and Electronic Fabrication
- Sealing & Potting
- Composites
- Heat Treat
- Adequate controls in place to ensure adequate review and control of process changes
- MAF processes
- Supplier processes
- Facilities and tooling processes

### STS-99/ET-92 Pre-Flight Review Action Item Process Change Control

- Current change system used for changes in process requirements is same system used for preparation/review/release of all engineering changes
- Process changes initiated by Technical Operations/Material Sciences
- Initiated as a result of in-house/supplier liaison calls, NASA direction, industry developments, etc.
- Proposed process change is prepared/reviewed by appropriate MAF disciplines
- Procurement (coordinates with Suppliers)
- **Production Operations**
- Quality/Safety

Environmental/Facilities

- **Technical Operations**
- Contracts
- Change release
- Contracts verifies authority for release, ensuring contract compliance
- change Release of engineering change concurrent with manufacturing process
- Released change forwarded to NASA Materials, Processes, and Manufacturing
- Process changes are included in readiness review process
- Not ET specific Change is briefed prior to a potential effectivity

## Supplier Process Change Control (Flowed Down to Supplier)

- Following process revision, Procurement Quality reviews specification to
- determine if changes are mandatory for Supplier
- If mandatory, Materiel Operations (Buyer) directs Supplier to incorporate change Supplier incorporates change and provides modified Process Instruction (PI) to Procurement Quality Control Field Representative for review
- Critical processes and PI violating LMMSS process requirements are routed to Buyer via Documentation Acceptance Sheet (DAS)
- Processes meeting LMMSS process requirements can be approved by Field Representative
- Buyer routes DAS/PI for review
- Product Technical Support
- **Material Sciences**
- Systems Engineering
- **Procurement Quality**
- If adequate, approved DAS/PI is routed back to supplier along with validation/re-validation requirements

## Supplier Process Change Control (Flow Up from Supplier)

- If Supplier desires to change PI, Supplier provides a modified PI to Procurement Quality Control Field Representative for review
- Field Representative coordinates with MAF Procurement Quality to determine appropriate action
- Activities similar to Process Requirements Flow Down
- e.g. DAS, validation/revalidation requirements, etc
- change) are required to be coordinated through LMMSS Other process changes (i.e. those which would not necessarily involve a PI
- LMMSS reviews changes for adequacy and provides validation/revalidation requirements
- For "sixth-buy" procurement, Suppliers are required to provide/maintain Process Flow Charts
- Process Flow Charts illustrate the sequence in which PI are used at Supplier
- LMMSS reviews changes for adequacy

# Facilities & Tooling (associated with ET manufacture) Change Control

- Operational Readiness Reviews are conducted to review facilities and tooling cnanges
- Mission Success/Safety reviews scope of change to identify what type of Operational Readiness Inspection (ORI) is warranted
- ORI verifies readiness of flight article, Facilities, Tooling, procedures, and personnel to perform specified operations
- ORI attendees include NASA representatives
- presented and reviewed in the readiness reviews Facilities and tooling changes associated with major ET design changes are

#### Summary

- and on-site operations and provide end-to-end oversight LMMSS change control management systems encompass supplier, sub-tier,
- Process changes are evaluated and reviewed to ensure successful **Implementation**
- Example 2 LMMSS systems and management practices are designed to ensure that no unplanned changes occur
- Process changes included in this readiness review are not ET specific
- Change is briefed prior to a potential effectivity

### **Critical Process Changes**

### B02042-001 and B02042-012

- Not implemented on ET-101
- Will be implemented on NCD for first use (NET ET-113)
- Use will be presented to/and approved by FCB and SMR

#### **Change Description**

- Release STP 5510 (Friction Plug Weld Repair) to incorporate a solid state weld repair process Revise STP 2506 (Ultrasonic Inspection of Weldments) to incorporate ultrasonic inspection of friction

Increase process reliability and weld repair strength

#### **Basis for Certification**

- SDS friction plug repair development program
- Fracture Control Board (MAF & MSFC) review/approval
- FPW repair mechanical property allowables
- Plug tensile
- Simulated service and surface crack tension
- Three point and tension fatigue
- Peaking and mismatch
- T and X intersection proof tests
- Heat repairs on plugs plug tensile
- SDS inspection technique development

#### Senior Management Review NCD To be presented in detail for first implementation as a

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## Weld Instruction Card (WIC) Certification

#### Issue

During a recent weld schedule review, it was determined that several of the weld schedules may have been incorrectly certified

#### Background

- 1 - 3 - 1

- Weld certification requirements are established through Engineering process requirements documents
- Certification requirements verify that weld schedules produce acceptable weld strengths and quality
- Requirements are then restated and implemented in the "how to" manufacturing process documents
- Ambiguity of the weld process documents and reliance on institutional knowledge led to instances of incomplete testing for weld certification

## Weld Instruction Card (WIC) Certification

#### Background

- Performed LMMSS Quality and Engineering review of all "as-built" ET welds
- Discrepancies were documented on non-conformance documents (NCDs) Each NCD was individually analyzed, dispositioned and approved by LMMSS and
- When necessary, additional welded panels were fabricated and tested
- All NCDs were dispositioned and approved All discrepant weld schedules were reviewed to the correct certification condition
- No safety of flight issues have been identified

### Actions Taken to Resolve Non-Conformances

- LMMSS and NASA/MSFC conducted a review (October present) of all ET welding
- Several findings were noted during the review that require resolution
- 0 Category 1 Findings: Require immediate implementation before welding can
- 19 Category 2 Findings: Require rapid implementation in accordance with a waterfall
- 30 Category 3 Findings: Would be implemented into weld operations on a non-

## Weld Instruction Card (WIC) Certification

## One Additional Issue Identified During the NASA/LMMSS Review

- Concern for design strength of welds (cryogenic properties) due to effect of weld parameter variations
- NASA/LMMSS Team reviewed the possible impact on ET-101

### ET-101 Rationale for Flight

- Team conducted a review of the following and determined that adequate rationale existed for ET-101 clearance:
- Weld-by-weld reviews
- Review of as-built parameter charts
- ET-101 as-delivered weld tests
- Parameter range test data
- Wide panel and confidence panel data
- Fracture property review
- Proof test stresses and history
- Flight stress analysis
- LMMSS and NASA/MSFC Review Team concluded that there was no safety of flight concerns

### WIC Issue Previously Presented To:

- Lead Center Program Managers Council on 09/10/99
- Daily PRCB (Mr. McMonagle) on 9/28/99
- Special Presentation to Mr. Dittemore on 10/19/99
- Program Managers' Preventive/Corrective Action Review on 11/05/99
- Final root cause will be presented to Chief Engineers Council

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## ET Electrical Wire Harness Investigation

#### **Actions Taken**

- External Tank Project conducted an in-depth review to ensure adequate safeguards and continued safe operation for flight
- The following items were reviewed:
- Harness design, fabrication and installation
- Final acceptance and checkout
- Post fabrication and installation harness disposition
- KSC inspection, checkout and acceptance
- ensure continued safe operation Investigation showed that ET cabling design, fabrication and installations have adequate safe guards to
- Some enhancements were identified

#### **Enhancements**

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Shield termination/splice improvement

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Meltable
Liners
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ECD 5/00

ECD 1/00

Solder Sleeves

Redesign of cable support tools to incorporate protective enclosure	<ul> <li>Re-evaluate following study and development</li> </ul>
ECD 06/0	ECD 1/01

Ţ Design of cable protection tool at KSC

Improved design (box-cushion) will be incorporated in next procurement

Long term action - P-clamp cushion configuration Complete ECD 1/04 00

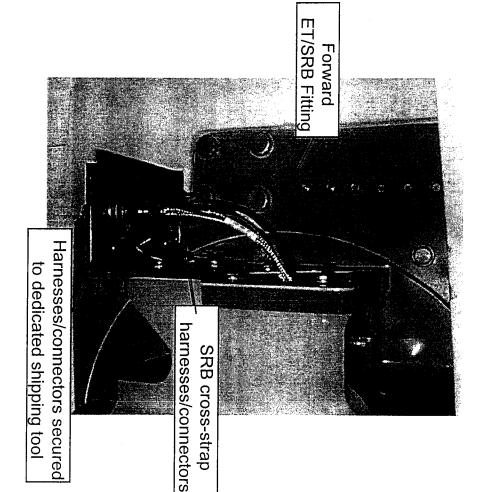
# ET/SRB Cross-Strap Harness Removal/Replacement

#### Issue

Damaged +Y ET/SRB cross-strap harness on STS-103/ET-101

#### Background

- Prior to forward crossover installation, procedure specifies performance of a premate inspection of cable connectors
- Inspection of STS-103/ET-101 revealed a damaged overall shield of +Y ET/SRB cross-strap harness
- Damage approximately 5 inches from face of connector
- Visual inspection indicated 3-5 broken shield strands and damaged wire identification tag
- Subsequent tactile inspection detected a damaged Kapton insulation/ nicked wire conductor that exposed bare wire
- Most probable cause was impact damage from portable scaffolding
- Harness assembly carries Crit 1R functions for SRB RSS



# ET/SRB Cross-Strap Harness Removal/Replacement

#### Discussion

- Decision made to remove and replace harness
- +Y and -Y harnesses are bundled together
- Easier to remove both harnesses rather than separate harnesses from each other

#### Rationale for Acceptance

- No design change required
- Replacement per existing engineering requirements
- lest
- DC continuity resistance of each ET/SRB cross strap harness
- DC isolation resistance of ETSRB cross-strap harness from left (-Y) connector contacts to right (+Y) connector contacts
- Inspection
- Inspection of cabling, connectors and clamps as part of Intertank egress
- Pre-mate inspection of cable connectors

### Effective tests/inspections in place to detect damaged hardware

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#### SRM&QA Assessment

as of 11/05/1999 Item

**ALERTS** 

MRB Assessment

**Hazard Assessment** 

### Corrective Action Problem Summaries (CAPS)

- T-067PF, Loss of Intertank SOFI during Ascent of STS-87/ET-89
- Documents IFA STS-87-T-001 actions

Status

None open

Complete - No issues/no impact to hazard

controls

Complete - All hazard reports were reassessed and hazard controls verified

Closed pending action status submitted to MSFC

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### **KSC Processing - Status**

as of 11/05/1999

# Discrepancy Reports (DRs)/Problem Reports (PRs) Review

- All PR/MRB discrepancies/dispositions similar to previously flown vehicles
- All MRBs have been reviewed by MAF Reliability for potential SMR

### Limited Life Component/ET Status

All within required life through scheduled launch date plus 90 days

### OMRSD Exceptions/Waivers: None

#### Removal/Replacement Activities

- Remove/replace composite nose cone
- Remove/replace ET/SRB cross-strap harness (Previously discussed as Special Topic)

#### **Simulations**

- Launch (S0044)
- Load (S0056)
- Mission Management Team
- Complete 10/18/1999
- Complete 10/15/1999
- 11/18/1999

### KSC Processing - Status as of 11/05/1999

#### **ET modifications**

- Mod Kits
- DC&R Bldg 45
- FEC

#### None

Remove/replace ET composite nose cone

- Intertank thrust panel foam reduction
- Intertank thrust panel foam venting
- Extend Intertank thrust panel foam venting
- Additional Intertank thrust panel foam venting and skin/stringer panel foam venting

No new items \*\*

None

No new items \*\*

Facility

Software

OMRSD/LCC

\*\* Previously presented for STS-92/ET-99 (Included in Appendix E)

#### Agenda

Overview

**Current Mission/External Tank Highlights** 

ET-101 Mass Properties Status

Changes

Special Topics

STS-99/ET-92 Pre-Flight Review Action Item

Critical Process Changes

Weld Instruction Card (WIC) Certification

ET Electrical Harness Investigation

ET/SRB Cross-Strap Harness Removal/ Replacement

Senior Management Review NCDs: None

SRM&QA Assessment

KSC Processing

Verification/Certification

Mission Unique Assessment

Readiness Statement

Appendices

A: Minor Changes

B: NCD Assessment Criteria

C: KSC Processing

Level II Exceptions/Waivers and Level III Deviations

E: Items previously reviewed for STS-99/ET-92

Jim Feeley

Rick Spring

Jim Feeley

Don Bolstad Leed Colon

Mike Bankester Juan Ramirez *Jim Feeley* 

### Verification/Certification

as of 11/05/1999

Item

### NSTS 07700, Volume X Baseline

Revision M, Change 254

#### Certification documentation

- Hardware Certification Sheets (HCS)
- Certificates of Qualification (COQ)

Waivers/Exceptions/Deviations

Status

No Issues

Complete - No Issues

No new items

#### as of 11/05/1999 Mission Unique Assessment

Category/Item

MPS Preflight Prediction Package Review of MRB actions using updated loads

**STS-101** 

ECD 11/19/1999

ECD 11/19/1999

#### Readiness Statement

completion/closure of open and planned work The External Tank, ET-101, is hereby certified and ready for STS-103 flight pending

Terry L. Hibbard
Vice President, External Tank Project
Lockheed Martin Michoud Space Systems

#### Appendices

A: Minor Changes

B: NCD Assessment Criteria

C: KSC Processing

PR/MRB Summaries

Modification Summaries/Status

Limited Life Component Status

Facility Modifications: No new items

Alternate Blowing Agent Usage at KSC

Software Changes: None

OMRSD Exceptions/Waivers: None

OMRSD Changes (RCNs): No new items

D: Level II Exceptions/Waivers and Level III Deviations

E: Items previously reviewed for STS-99/ET-92

## Change Presentation Criteria

- Changes for Presentation
- Type I Documentation
- Requirements
- Configuration (affects part interchangeability)
- Materials (requires SE16 revision)
- Technical Changes to Processes (STPs)
- Changes for inclusion in Appendix
- Minor configuration (affects part interchangeability)
- Minor updates Does not affect part interchangeability, materials or processes
- Revise point of incorporation
- Drawing clarification
- Non-flight hardware installation/removal
- Datum changes
- Update usage/List of Materials
- Drawing incorporation errors
- Manufacturing aid
- Drawing errors
- Fastener type/grip
- Producibility
- Update vendor list
- Laboratory test procedures
- Materials
- All Class I and Class II changes have been reviewed by the MSFC Resident Office
- and Manufacturing Department for concurrence All Class II STP/STM/Standard/Code sheet changes have been submitted to the MSFC Materials, Processes

## Minor Configuration Changes

#### Description

#### Deletion of ET Range Safety Systems (B02010-047/-029)

#### Increase base thickness of aft Orbiter fitting to eliminate custom shim (H35809-459)

#### Reason

- Deleted residual isolators and supports associated RSS installation
- Reduction in web thickness on SLWT from LWT caused aft orbiter fitting installation problems
   Custom shim was an interim fix
- Custom shim was an interim funtil fitting redesign was implemented

#### **Basis for Certification**

- Similarity
- > ! !-
- Analysis
- Similarity

## Minor Changes – Structure

Change No.

Description

B01982-B49 B02022-046	Increased material stock thickness by .005in because parts thinned down more than expected during stretch forming  Update SLWT Weld Acceptance Manual to incorporate results of allowables testing
H35809-182	Corrected location of datum on LH2 cable tray support electronic drawing. Error during conversion from manual to electronic drawing
H35809-245	Clarified note on fairing assembly drawing to apply epoxy on non-adhesive side of tape
H35809-363	Revise vertical strut forging drawing to allow for the previously approved penetrant inspection Method D (post emulsified)
H35809-377	Deleted ultrasonic inspection requirement from outer chords due to thin chord material
H35809-404	Revise lower chord and bottom support chamfer drawing to better define chamfer and radius dimensions
H35809-457	Corrected cable tray support assembly optional washer from AN960PD416 to AN960KD416
H35809-469/ -478	Revised machined fitting drawing to agree with manual drawing. Error during conversion from manual to electronic drawing

## Minor Changes – Structure

Change No.

Description

H35809-470	Removed dimension on LO2 tank and LH2 forward dome cap. Added note for depth of cut at weld land transition. Tolerance relaxation for producibility enhancement
H35809-544	Add nutplate holes sizes to Left and Right vertical strut to cable tray fairing drawing. Error during conversion from manual to electronic drawing
H35809-561	Clarified/added dimensions missing from machined thrust fitting drawing. Error during conversion from manual to electronic drawing
H35809-769	Revise dimension of forward level probe assembly from 3.210 max. to 3.220 max. to preclude NCDs. Part previously machined at detail level – Now machined at assembly level
H35809-791	Replace obsolete DA-15 (MA76A) sealant with a previously approved sealant Type III silicone paste (M660C)
H35809-814	Revised note to place tolerances in scribe lines for manufacturing use in aligning LO2 barrel panel welded assembly
H35809-854	Revise machined fairing assembly dimension (.42 constant to .42 +.06/03) to allow new supplier to use numerically controlled program to build parts
H35809-859	Revise Frame 1973 to barrel 1 attach fittings tapped hole size and machined fitting end pad from 0.25" min. to 0.30" min.

## Minor Changes – Structure

Change No.	Description
H35809-886	Add STP 2003 reference to drawing in addition to 100 degree countersunk fastener requirements
H35809-929	Revised view to extend vibro mark line past the theoretical weld line. Prevents rework to extend line if trim prior to weld removes entire line
H36809-233	Increased extrusion width due to not being able to meet minimum requirement after forming
H36809-253	Revised height dimension on drawing to include web thickness (dimension unchanged)
H36809-255	Corrected sheet numbers to eliminate duplicate sheet numbers
H36809-257	Clarify drawing to specify that no primer was required on part. Hardware built to requirements of this change
H36809-290	Revised drawings to reflect dimensional tolerances of new supplier of attach point castings
H36809-292	Added missing radius dimension. Error during conversion from manual to electronic drawing

# Minor Changes – Propulsion/Mechanical

Change No.	Description
J31162-003,	Reformatted tube drawings, created new development tube part numbers with the purpose
-004,-005,	of developing/incorporating new bend date to eliminate NCDs for interference on old tubes.
-006,-007,	No requirement, design, material or fitting
-008,-009,	

H35809-822/-824 Intertank pneumatic tubing installation drawing was clarified to update/clarify/reference correct details and change two screw length to make hardware installation consistent

-010,-014, -016,-017, -018,-019, -020,-024

H36809-239 Revise MAF surface acceptance criteria document to be consistent with supplier document

### Minor Changes - TPS

Change No.

**Description** 

H36809-205 H35809-962 Revised drawing to reinstate TPS sectional views that were inadvertently deleted by previous change

Revised MAF and Offsite Build Acceptance Criteria Manual to: Add Inconel acceptance criteria and Kapton tape section

Delete foam materials no longer used

Reinstate "Part B" in-process rework requirements to Offsite manual

Minor documentation/format revisions

H36809-273 Corrected drawing sheet to reflect appropriate revision levels of graphic sheets

### Minor Changes – Electrical

Change No. **Description** 

H36809-028 Revise Bill of Materials to specify correct RFI backshell. Correct RFI backshell that was

called out on drawing notes was used

H50001-803/-804 Replaced copper conductor wire with soft copper conductor wire for ease of manufacturing

### Minor Changes - Systems

Change No.

Description

H36809-296	B02024A-100	B02059-009 H36809-309	B02041A-033	B02024A-099 B02028-013
Revised ET Protection Requirements and Removal drawing to reflect the anti-static film requirement. No change to ET hardware - Documentation only	Release ALWT Thermal Data Book	<ul> <li>Revised effectivities and added inspections due to Intertank Foam Venting design</li> <li>Correct miscellaneous technical and non-technical errors</li> </ul>	<ul> <li>Revised effectivities and added inspections due to GUCA and ET GH2 Vent Disconnect redesign</li> </ul>	Updates to CIL Implementation Drawing  Revised effectivities and added inspections due to ALWT Prime Revised effectivities and added inspections due to Fairing Support Bracket design change

# Minor Changes – Material Sciences

### Change No. Document Description

- H36809-314 **EQTP 1003** Delete obsolete materials (NCFI 22-65, CPR 488, PDL 4034, BX-250 Type
- Delete GPC Test of Component A and replaced requirement with more accurate amine equivalent test
- and use GC technique for improved accuracy Change requirement for FTIR blowing agent peaks to "For information only"

## MRB Assessment Criteria

- Assessment of all MRB Items
- Adequacy of technical dispositions
- Completeness of required repair/retest
- First time occurrences/SMR
- Verification of margin of safety assessment/documented
- Review of impact to hazards

Engineering review for impact to loads

Engineering review for reduced extrusion to plate weld allowables

# ET-101 DR/PR/OMRSD Waiver/Exceptions as of 10/27/99

System	Discrepancy	Problem	MRB Items
• Electrical	0		0
<ul><li>Propulsion/</li><li>Pneumatics</li></ul>		ω	0
<ul> <li>Structures</li> </ul>	0	7	_
• Therm al (TPS)	0	1 2	ъ
Total		23	6

### ET-101 PR/MRBs, Structural

#### PR No./Status

ET-101-ST-0004

#### Title/Description

#### Disposition

alignment holes between the During nose cone removal/ horn sections louver spacers and the vent replacement noted suspect the

Use as repaired New pilot holes drilled to compensate for differences replacement spacer between original spacer and

### ET-101 PR/MRBs, Thermal

PR No./Status FT-101-TS-0004	Title/Description	Disposition
ET-101-TS-0004	Void in the +Y bulkhead	<ul> <li>Repair with PDL</li> </ul>

conathane bondline

#### Extended conathane cure at XT371 interface exceeded Humidity controlled until foam applied

ET-101-TS-0005

ET-101-TS-0006 Low foam noted during sanding operations

humidity requirements

- Use as repaired

   Areas were blended

   Thermal properties not affected
- Inadvertent venting of PDL

ET-101-TS-0007

- Use as is
- Testing demonstrates physical properties of foam not affected by venting
- Vented PDL flown on STS-96/ ET-100

C-3

### ET-101 PR/MRBs, Thermal

#### PR No./Status

ET-101-TS-0009

#### Title/Description

#### No PDL applied to aft fairing bolt hole

#### of aft fairing closeout underfills at forward edge Two areas of foam

#### Disposition

- Bolt hole closed out with PDL
- Additional Conathane bondline acceptable for flight
- Areas repaired using PDL
- Additional Conathane bondline acceptable for flight

# ET-101 KSC Modification Summary/Status

Modification No.	Description	Status
FEC KET-0048R1	Intertank thrust panel foam thickness reduction	Complete - 06/17/99
FEC KET-0054	Intertank thrust panel foam venting	Complete - 08/1/99
FEC KET-0054R1	Extend Intertank thrust panel foam venting	Complete - 08/19/99
FEC KET-0056	Intertank skin/stringer panel foam venting	Complete - 09/03/99

# ET-101 Limited Life Components

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Description	Part No.	Serial No.	• • •	Required	Remaining
LH2 Manhole Cover Assy	80914081488-509	0000650340		<b>→</b>	14
LO2 Tank Mechanical Assy	80912002000-500	0000691540	14	<b>&gt;</b>	14
LH2 Tank Proof Test	80914002000-500	0000676100	14	<u> </u>	14
LO2 Manhole Cover Assy	80911001205-509	0000696740	14	<u>~</u>	14
LO2 Fwd Ogive Cover Plate	80911001207-509	0000694300	14	<u> </u>	14
LH2 Manhole Cover Assy  – Aft	80911001444-509	0000684710	14	<u> </u>	14
LH2 Tank Cap Closure Siphon	80911001445-502	0000717100	14		14

# ET-101 Limited Life Components Age Sensitive

External Tank 80901000000-500 Description Part No.

Serial No.

00000094

6 Yrs

11/98

Limit

Initiation

Expiration

11/2004

# **ET-101 Limited Life Components**

Age Sensitive Pyrotechnics - 10 Years Life/Requires Lot Verification Tests After 4 and 7 Years

Description	Part No.	Lot No.	Lot No. Serial No.	Retest Retest	Expiration
Separator Assy GUCP	PD5000020-059 AAH	ААН	0001145	0001145 01/27/1998	01/2002
Cartridge Assy	PD5000020-030 AAH	AAH	0001292	01/17/1998	01/2002
Cartridge Assy	PD5000020-030 AAH	AAH	0001293	0001293 01/17/1998	01/2002

# ET-101 Alternate Blowing Agent Usage at KSC

PDL-1034 T5244	PDL-1034 T5249	SS-1171	Foam
T5244	T5249	T5141	Document
ET/Orbiter jack pad closeout	ET/SRB aft fairing closeout	Aft hardpoint closeout	Description
Xt 1129, 15/345 degrees	Xt 2058, 70/290 degrees	Xt 1973-2076, 180 degrees	Location
9 sq. in.	240 sq. in.	4410 sq. in.	Size

### SSP Waivers - Baselir

Document / Requirement	Authority	Paragraph	Waiver No.	Effectivity
NSTS 07700, Volume V - Information Management Requirements	formation Ma	anagement Req	uirements	
Generic part name, type, common designation, and part manufacturer H4 code are not required for EEE parts status print out	S00127B 06/28/88	Table C.5 Item 2SR-10	on.	ETs 49 & up
Data required in limited life status report is supplied in acceptance data package	S00127B 06/28/88	Table C.6 Item 2SR-7	œ	ETs 49 & up
Requirement to deliver element pressure vessel historical data to EPO will remain at contractor facility	S00127B 06/28/88	Table C.6 Item 2SR-7	10	ETs 49 & up

### SSP Waivers - Baseline

Document / Requirement	Authority	Paragraph	Waiver No.	<b>Effectivity</b>
NSTS 07700, Volume X – Space Shuttle Flight and Ground System Specification, Book 1 Requirements	Ground System	າ Specificatio	า, Book 1 Rec	luirements
Electrical wiring of redundant systems NSTS 08080-1, 20A/4B	S040732G 06/18/88	Table 2.0	362	ETs 23, 27-29, 31 & up
Soldering, circuit boards, and connectors	S040732M 09/09/88	3.6.15.1.1	379	ETs 23, 27-29, 31 & up
PD7400106 circuit boards	S040732M 09/09/88	3.6.15.2.1	382	ETs 23, 27-29, 31 & up
Pressure transducer circuit boards	S040732M 09/09/88	3.6.15.2.1	383	ETs 23, 27-29, 31 & up

### SSP Waivers - Baseline

Debris prevention	NSTS 07700 Volume X – Space Shuttle Flight and Ground System Specification, B	Document / Requirement Authority
S082962A 10/02/98	d Ground Syster	nt Authority
3.2.1.2.14	m Specification	Paragraph Wa
675	າ, Book 1 Requirements	Waiver No.
ETs 92, 97-105	uirements	Effectivity

SL-E-0001 – Specifica	of ET components
SL-E-0001 – Specification Electromagnetic Compatibility	03/07/89
Requirement	31 & u

Environmental acceptance

S002130H

applicable to the ET	Wire marking is not
09/27/89	S40732K
	3.2.5
	ယ
31 & up	ETs 23, 27-29,

SW-E-0002 – Ground Support Equipment General Design Re
upport Equipment
General Design
Requirements

to yield for GUCP leg	Factor of safety applied
08/17/88	S87020L
	3.4.2.1
	21
31 & up	ETs 23, 27- 29,

## SSP Exceptions - Baseline

**Document / Requirement** Authority Paragraph **Exception No.** 

NSTS 08123 Certification of Flexible Hoses and Bellows

LH2 feedline bellows

S005203G-R1 08/02/91

ယ

ETs 42-45, 47

dn %

**Effectivity** 

D 4

## SSP Deviations - Baseline

Document / Requirement

Paragraph Deviation No.

**Effectivity** 

# CPTO1M09A, External Tank Contract End Item Specification

Photographic evidence shows a loss of foam from both +Y and –Y Intertank Thrust Panels

3.2.5

233

ETs 92, 97- 99, 101-105

Analysis concluded a low probability of debris impact on Orbiter critical area

#### **Pad Facility Modification**

LO2 pump controller replacement

- **OMRSD** Delete ET Intertank inspections and revise leak checks at KSC (KT13565A)
- Revise GUCP quick disconnect leak check methods (KT13576M1)

Revise tank pressure monitoring requirement and increase tank pressure

lower limit (KT13566M)

- LO2 Orbiter/ET carrier plate purge pressure (KS13664M)
- Tank pressure monitoring requirements (MT13625M/MS13626M)

#### Minor Changes

### **Pad Facility Modification**

- SSP58891 LO2 Pump Controller Replacement
- LO2 pump Robicon variable frequency drive (VFD) controllers were replaced with Allen Bradley VFDs
- Installation and validation completed 7/2/99

# Intertank Door Closeout OMRSD Changes, File IV

#### File IV

- Background
- KSC Launch Flow Enhancement team requested Intertank door closeout prior to shipment to KSC
- Results in flow savings of 5 serial days of ET standalone processing
- MAF engineering revised to closeout Intertank door at MAF for ETs 93-95 and ET-106 and subs (B02049)
- Several KSC inspections and leak checks would be deleted/revised as result of door closeout
- KSC requested earlier deletion of inspection/leak checks to realize savings beginning with ET-92

# ntertank Door Closeout OMRSD Changes, File IV

- Description of Change
- Delete ET Intertank inspections and revise leak checks at KSC (KT13565A)
- Changed inspections (nose cone and Intertank purge tube GH2 vent valve sense line flow test to contingency to be electrical cabling, LH2 ullage pressure transducers) and internal

performed only when Intertank access has been established

- Replaced leak checks that are performed inside the Intertank tank feedthroughs) with leak checks performed external to the Intertank (LO2 and LH2 tank pressure ports, GH2 pressurization line, LH2
- Rationale for Acceptance
- Similarity Intertank inspections and leak checks are performed at MAF prior to shipment to KSC
- Approved by Level II PRCBD on 2/11/99 (S053293EJ)

# Intertank Door Closeout OMRSD Changes, File IV

- Description of Change
- Revise tank pressure monitoring requirement and increase tank pressure lower limit (KT13566M)
- Extended tank pressure monitoring interval from once per week to within 14 calendar days of last reading as long as pressures
- are within specification during tank move operations Intertank door closeout eliminates continuous monitoring of tanks during Orbiter mate
- additional assurance that tank pressure will not decay below Increased tank pressure lower limit from 1.7 to 2.2 psig for mınımum requirement
- Rationale for Acceptance
- Adequate tank pressure is maintained
- Historical pressures have not varied +/- 0.5 psig during VAB Integration Cell processing
- Approved by Level II PRCBD on 2/11/99 (S053293EK)

# Intertank Door Closeout OMRSD Changes, File IV

- Description of Change
- Revise GUCP quick disconnect leak check methods (KT13576M1)
- Change GUCP 3/8" quick disconnect leak checks to a nonot using a flow meter in the VAB intrusive method using the mass spectrometer at the pad instead
- GO2/GH2 vent valve, Helium inject, HGDS, Nose cone purge and Intertank purge
- Acceptable leak rates were adjusted due to the sensitivity of the measuring device
- Rationale for Acceptance
- Similarity Revised leak check methods are adequate to detect leakage at the GUCP quick disconnects
- Approved by Level II PRCBD on 2/11/99 (S053293EM)

### LO2 Orbiter/ET Carrier Plate Purge Pressure

#### KS13664M

- Background
- Purge pressure is required to establish plate gap pressure of 0.25 to 0.7 psig
- Supply pressure for LO2 ET/Orbiter disconnect purge was below OMRSD limit of 375 psig during set-up on STS-95 (ET-98) and STS-96 (ET-100)
- Waivers approved that accepted lower supply pressure based on analysis that indicated a tight system
- Description of Change
- Revise minimum GSE supply pressure for LO2 Orbiter/ET carrier plate purge
- Was: 375 psig
- Now: 300 psig
- Rationale for Acceptance
- Analysis
- Lower set point pressure protects for adequate plate gap purge flow
- Provides for dilution of an aft compartment LCC leak to below 25% O2 concentration with a factor of 4
- 300 psig provides adequate protection against plate gap
- Approved by Level II PRCBD on 7/16/99 (S053293KW)

### Tank Pressure Monitoring Requirements File IV and II, MT13625M/MS13626M

- Background
- ETs previously shipped to KSC with psig tank pressure monitoring gages ınstalled
- Calibration inaccuracies in psig gages caused change to psia gages on ETs for shipment to KSC
- After ET offload to the KSC VAB, the ET is monitored by KSC pressure monitoring system using psig gages
- Description of Change
- Added tank pressure requirements in psia values for monitoring periods prior to gage installation in the KSC VAB
- Tank pressure requirements 2.2-9.7 psig 9R 16.9-24.4 psia
- Rationale for Acceptance
- Similarity Positive tank pressure maintained
- Approved by Level II PRCBD on 7/28/99 (S053293HH and S053293HF)

### Minor Class II Changes

Change No. **Description** 

Structural

H34809-986 Added note on slosh baffle frame fitting to define number in brackets called out on the

field of the drawing

H35809-886, 887, 888, 889 Corrected drawings missing reference to STP2003, countersink and flushness

requirements

H35809-972 Revised primer application on upper aft ET/SRB fitting to touch up unprimed areas

H36809-059 Revised Thermal Isolator Assembly drawing to assure proper application of J536 tape;

issued rework DCN to apply tape to side opposite the assembly part number marking

Deleted references to spotface areas on the Intertank to LH2 Splice Installation; there are

not spotface areas on drawing

H36809-060

**Propulsion** 

H36809-072 Added one fastener to list of materials

through -005 J31196-001 requirement. Shortened tube lengths, created new spacer and revised tube bend data. Revised helium inject installation to allow flex hose to meet the minimum bend radius

### Minor Class II Changes

Change No. Description

**TPS** 

H35809-983	Added foam void acceptance criteria to LO2 Feedline support rib locations.
H35809-985	Added application of BX-250 per STP 1503 as an alternative to application of SS-1171 per STP 1536 for manual spray application process on LO2 dome
H35809-986	Revised minimum/maximum foam thickness at component build of the flex LO2 feedline to agree with venting inspections done at installation.
H36809-019	Added requirements to control the gap between Composite Nose Cone Seal (MA-25 ablator) and nose cone bulkhead brackets to prevent erosion of NCFI 24-124.
H36809-066	Revised LO2 PAL ramp configuration to eliminate less than adequate foam coverage over intertank stringer S7-3. This condition resulted from range safety deletion.
H36809-158	Clarify referenced dimension point of origin on LH2 tank flange at Sta. 1123.15.
H50001-799	Foam migration into adjacent acreage foam rind at Intertank flange foam closeouts; added optional foam closeout at Intertank stringer end

### Minor Class II Changes

Change No. **Description** 

H50001-800 Relax flushness requirement on LO2 tank plug pulls and to allow foam migration from

plug pull repairs to crevices in acreage foam.

H50001-801 Revised LH2 Tank third hard point trim to allow for foam shadowing

TPS

H50001-802 Revised LH2 aft dome cryoflex test panel to waive epoxy primer touch up not required

out side test area

Electrical

H50001-808 Replaced hard copper conductor wire with soft copper conductor wire for ease for

manufacturing

Systems

H36809-098 Clarified the type of shoe soles allowed Walking Loads Limitations document. Added

drawing references, tools and associated graphics. Added note to specify limit on ladder

#### Minor Class II Changes

Change No.

**Description** 

H36809-220 measure the LH2 and LO2 tanks. Removed tool locations and numbers. Incorporated Revised ET measurement requirements to remove all references specifying how to

other minor corrections references.

Minor Class II Changes

Change No. Description

Systems

H36809-262 such as SOFI/Ablator dust and foam mist overspray particles produced during VAB Revised Acceptance Criteria documents to specify that atmospheric fallout type elements

acreage spray process are acceptable.

H36809-238 Revised Loads Data Book to include current database history for LWT/SLWT design and

update specific components.

H36809-796 Revised ET Protection Requirements to: identify installation point for GO2 press. line and LO2 feedline fairing; delete protective wrap on bipod fittings and protective tools from

nose cone cable tray fairing and louvers; revise GH2 vent line bellows cover to reflect as-

built processing

J31176-011 & mechanical assembly and lug weld and LH2 aft dome cap drawings Clarified ICD requirements on drawings; added reference to ICD 2-00001 on LO2

### Minor Class II Changes - STPs/STMs/STDs/AVL

Change No. Document

Description

H36809-202 STP 6514 - \			H36809-190 1118 - Alumin and Sheet		H36809-167 STP6001 Adhesive Bonding
STP 6514 – Wire Bonding to LH2 Interior			1118 - Aluminum Alloy Plate and Sheet		ding ' ::
Removed optional cleaning material (Freon PCA) from list of materials. Isopropyl Alcohol (IPA) already listed as approved	<ul> <li>Added AMS-QQ-A-250/30 as alternate spec for material composition/properties</li> <li>No change to material used</li> </ul>	<ul> <li>Added clarifying note requiring STM1701 for thickness range of .250 to .499 inch</li> </ul>	<ul> <li>Moved callout for ultrasonic inspection per MIL-STD-2154</li> <li>Class B for plate exceeding 2000 lbs from drawing to material code sheet</li> </ul>	<ul> <li>Updated adhesive name from CREST 7450 to LORD 7450 to reflect vendor name change</li> <li>No change to processes or chemicals used</li> </ul>	<ul> <li>Revised to refer to solvent wiping per STP5002 rather than specifying cleaning instructions within the bonding STP</li> </ul>

### Minor Class II Changes - STPs/STMs/STDs/AVL

Change No. Document

**Description** 

B02041A- 015	H36809-261	H36809-243
62A1 Stainless Steel Spring Wire	N308 Drylube Lubricant	Y498 – Nitrogen Propellant & Replaced "Helium" inad Press Agent "Nitrogen"
Released material code sheet for Type 302, Class 1 stainless steel spring wire; procurement specification ASTM A313	Revised code sheet to specify that one can per lot be shipped to MSFC for LO2 Compatibility Testing specimen preparation	Replaced "Helium" inadvertently placed in code title with "Nitrogen"

		,